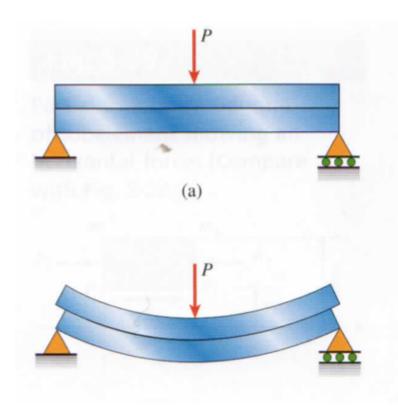


"You have clearly been under enormous stress."

Lecture 12: Shear stresses in beams

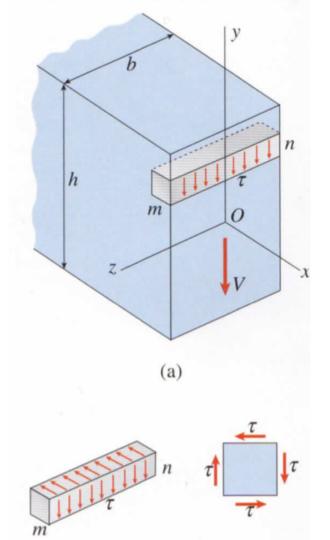


Shear stress in beams

We've seen that if a beam is in pure bending, the only stresses that act on the cross section are normal stresses

In non-uniform bending, we will have normal stresses and shear stresses

Assume two identical beams in bending. Under load, one beam will slide over the other. The force that is required to prevent this sliding is the longitudinal shear force.



For the case of a rectangular cross-section we can assume:

- Shear stresses that act on a cross section are evenly distributed from one side to the other and act vertically
- Shear stresses acting on one side are accompanied by shear stresses of equal magnitude on a perpendicular face
- Therefore we will have vertical AND horizontal shear stresses

First result: at the top and at the bottom of the beam, the horizontal shear stresses are zero. Therefore also the vertical shear stresses are zero.

NON-UNIFORM BENDWS =) M.V

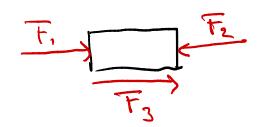
A START WITH WORMAL STRESSES

 $\frac{1}{2}$

NORMAL STRESSES ON MM: $G(Z) = \frac{M \cdot Z}{T}$ NORMAL STRESSES ON MM: $G(Z) = \frac{h \cdot dh}{T} \cdot Z$



m m



IF THE MOMENT@ mp + mp => 2Fx +0

=) FOR SIFX = 0 => THERE MUST BE

A SHEAR FORCE

$$F_{1} = \int_{2}^{N_{2}} \sigma_{1} dA = \int_{2}^{N_{2}} \frac{M_{2}}{L} dA$$

$$F_{2} = \int_{2}^{N_{2}} \sigma_{2} dA = \int_{2}^{N_{2}} \frac{(M_{1} + dM)}{L} dA$$

$$F_{3} = \int_{2}^{N_{2}} \sigma_{3} dA = \int_{2}^{N_{2}} \frac{(M_{1} + dM)}{L} dA$$

$$T_1 + T_2 = T_2$$
 => $T_3 = T_2 - T_1$

$$\overline{T_3} = \int \frac{M \cdot dM}{T} z dA - \int \frac{M}{T} z dA = \int \frac{dM}{T} z dA = \frac{dM}{T} \int z dA$$

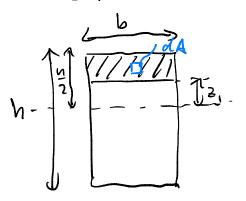
ME-231B / STRUCTURAL MECHANICS FOR S

SINCE WE ASSUMED THAT O IS EQUALLY DISTRIBUTED OVER WIGH

$$F_3 = 2 \cdot A = 2 \cdot b \cdot dx = 32 = \frac{r_3}{b \cdot dx}$$

$$\mathcal{Z} = \frac{dM}{dx} \cdot \frac{1}{Ib} \int z dA$$

$$= \sum_{S^{\perp}} \sum_{S^{\perp}} \frac{1}{S^{\perp}} \frac{1}{S$$



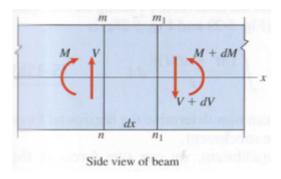
$$Q = \int_{2}^{2} dA = \int_{3}^{2} b \cdot 2 d2 = \frac{b}{2} \left(\frac{h^{2}}{4} - Z_{1}^{2} \right)$$

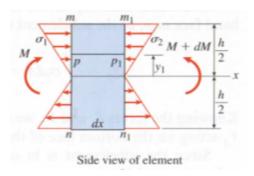
WITH SHEAR FORMULA WE GET:

$$C = C(2) = \frac{\sqrt{h^2 - 2}}{2I} \left(\frac{h^2 - 2}{4} - 2 \right)$$

THAX =
$$\frac{3}{2}\frac{V}{A}$$
 => THAX IS 50% LARGOR THAN AVERAGE SHEAR STRESS.

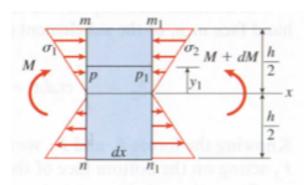




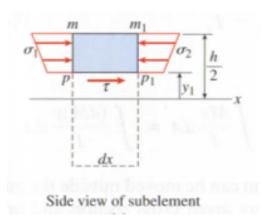


- Assume a beam in non-uniform bending and take two vertical cross-sections a distance dx apart.
- Since we have non-uniform bending, we have both normal stresses and shear stresses.
- We want to find an expression that relates external forces to shear stresses
- Our approach:
 - Use the fact that shear stresses act both normal to the beam axis as well as parallel to the beam axis
 - Use the expression for normal stresses (flexure formula) to calculate forceequilibrium in the longitudinal direction in the presence of a moment gradient.

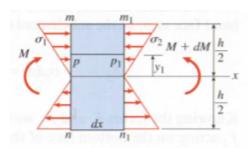




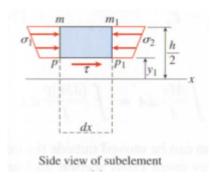
Side view of element



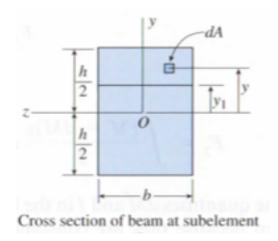
- We calculate the normal stresses on each cross-section of the element using the flexure formula
- The stresses vary linearly with y, being zero at y=0
- We know that shear stresses are stresses that occur between two sides of a cross-section. We make a horizontal cut through the element to look at the sub-element mm₁pp₁.
- Calculate the forces acting on the subelement in the x-direction.



Side view of element



- If the bending is uniform, then there is no difference in moment on the face mp and the face m₁p₁. Therefore the normal forces cancel each other out.
- In non-uniform bending, there is a nonzero dM between the two cut planes.
- This means that $\sigma_1 \neq \sigma_2$. Which results in a force difference on the two faces.



- We calculate the forces acting on the faces mp and m₁p₁ by integrating over the crosssectional area of the sub-element.
- Since we are interested in the shear force at a specific value for y=y₁, we calculate the integral over the area of b*y₁ to b*h/2
- Due to the need for equilibrium in the x-direction, the difference between the two forces at the two sides of the sub-element has to be compensated by an additional longitudinal force, which is the shear force between two horizontal slices through the beam.

From the in-plane shear force we can then calculate the shear stress and find the shear formula:

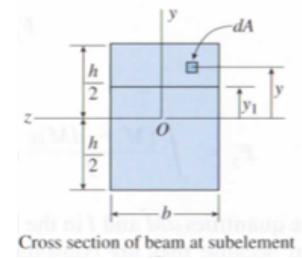
$$\tau = \frac{V \cdot Q}{I \cdot b}$$

with Q being the first moment of area:

$$Q := \int_A y \cdot dA$$

 The last thing we need to do to describe the shear stress distribution in the beam is to calculate Q.

- For a beam with a rectangular crosssection, we can calculate $Q(y_1)$ by calculating the area from y_1 up to h/2
- We could also calculate the area from y₁ downward, and would bet –Q
- We can easily show that at $y_1=0$, Q is maximum and at $y_1=+/-h/2$, Q=0
- Through integration and combination with the shear formula we get:



$$\tau(y) = \frac{V}{2I} \left(\frac{h^2}{4} - y^2 \right)$$

Shear stress in beams

$$\tau(y) = \frac{V}{2I} \left(\frac{h^2}{4} - y^2 \right)$$

We see now:

- shear stress varies quadratically with the distance from the neutral axis
- shear stress is zero at the beams upper and lower surfaces
- the shear stress is maximum at its neutral axis:

$$\tau_{max} = \frac{Vh^2}{8I} = \frac{3V}{2A}$$

 the maximum shear stress is 50% higher than the average shear stress

Validity of the shear formula

Georg Fantner

During the derivation of the shear (and flexure) formula we have made a number of assumptions to make the derivation easier.



It is important to use the formula only in cases where these assumptions are justified:

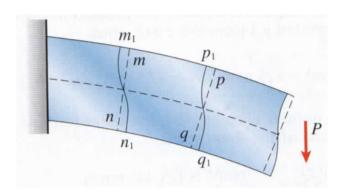
Edges of the cross-section must be parallel to the y axis

The shear stresses must be uniform across the width of the cross-section

The beam must be prismatic (e.g. must have a constant cross-section). The formula is not correct for a tapered beam.

Effect of shear stress

- \blacksquare The shear stress distribution in the beam cross section is quadratic in the plane of bending. $_{\tau}$
- From Hooke's law in shear we know: $\gamma = \frac{\dot{r}}{G}$
- Therefore also the shear strains are not constant along the cross-section. This means that plane cross-sections that were initially normal to the beam axis, will no longer be plane after bending!





Calculating shear and moment diagrams

